SHUTTLE CRITICAL ITEMS LIST - ORBITER

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FUNC:

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SUBSYSTEM : LANDING DECELERATION FMEA NO 02-1F -G09-CC-J01 REV:06/27/88

ASSEMBLY :MLG STRUT ACTUATOR

/N RI :MC287-0034

P/N VENDOR: PARKER-BERTEA

QUANTITY : 2

:ONE PER ACTUATOR

VEHICLE

PHASE(S): PL

CRIT CRIT. HDW:

103 104

102 EFFECTIVITY: Х X

X LO 00 DO X LS

PREPARED BY:

APPROVED BY: DES

REDUNDANCY SCREEN: A-PASS B-FAIL C-PASS

APPROVED BY (NASA): SSM R. Balcimas 8

DES REL

N LEVERT C NELSON

REL CA QE 015 8. Comin 7-25-88 REL 160 ch C + - 5/17/50

QE

M SAVALA

ITEM:

CHECK VALVE, CAVITATION

FUNCTION:

PERMITS FLUID FROM SUPPLY TO ENTER THE PISTON END OF THE CYLINDER DURI: GEAR EXTENSION PHASE TO PREVENT CAVITATION. PREVENTS REVERSE FLO (BACKWASHING) THROUGH THE EXTEND PORT FILTER DURING GEAR RETRACT PHASE AND MAINTAINS FLUID IN THE ACTUATOR IN THE EVENT OF UPSTREAM LI FAILURE.

FAILURE MODE: FAILS OPEN

AUSE(\$):

CONTAMINATION, BROKEN SPRING

EFFECT(S) ON:

(A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE

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- (A) AT DOWN GEAR COMMAND LOSS OF ACTUATOR DAMPING FUNCTION.
- (B) POTENTIAL FOR LANDING GEAR DAMAGE DUE TO EXCESSIVE DEPLOYME: VELOCITY. া প্র

10 July 2007

- (C,D) POTENTIAL FOR LOSS OF CREW/VEHICLE WITH TWO FAILURES: RUPTURE (RETURN LINE BETWEEN ACTUATOR AND RETURN LINE CHECK VALVE AND ACTUATO CHECK VALVE FAILING OPEN, PERMITTING EXCESSIVE VELOCITY OF GED DEPLOYMENT WHICH MAY RESULT IN DAMAGE/LOSS OF THE LANDING GEAR.
- (E) FUNCTIONAL CRITICALITY EFFECTS-SEE ITEM (D) ABOVE. "B" SCREEN : FAILED BECAUSE THERE IS NO INFLIGHT INSTRUMENTATION AND THIS FAILUR WOULD ONLY MANIFEST ITSELF UNDER A RUPTURED LINE CONDITION.

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SUBSYSTEM : LANDING DECELERATION FMEA NO 02-1F -G09-CC-J01 REV:06/27/

DISPOSITION & RATIONALE:

(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN

ANTICAVITATION ORIFICE UPSTREAM OF CHECK VALVE HAS AN 80 MICRON FILTER EACH END. SYSTEM CLEANLINESS IS TO LEVEL 220 PER MA0110-301. EXTEND RETRACT PORTS ARE PROTECTED WITH AN 80 MICRON FILTER. CHECK VA INCORPORATES INNER AND OUTER (DUAL) NESTED SPRINGS. THE SPRINGS ARE M FROM CORROSION RESISTANT MATERIAL (17-7PH) WITH MODERATE STRESS AND GUIDED.

(B) TEST

QUALIFICATION-RANDOM VIBRATION 0.4 G2/HZ, 12 MINUTES AT EACH AXIS 0.14 G2/HZ, 34 MINUTES AT EACH AXIS. ENDURANCE TESTS, 100 DUTY CYCLES EACH TEMPERATURE: -35, 20, 90 AND 140 DEGREES F. ONE DUTY CYCLE EQU 10 DEPLOY/RETRACT SEQUENCE. ACTUATOR EXTEND TIME TEST, PROOF PRESS TEST, PERFORMANCE RECORD TEST INCLUDING HIGH PRESSURE STATIC EXTER LEAKAGE TEST, LOW PRESSURE STATIC EXTERNAL LEAKAGE TEST AND DYNAMIC S LEAKAGE TEST. POST TEST PROCEDURE INCLUDES DISASSEMBLY AND INSPECT OF WORKING COMPONENTS. 400 FULL STROKE CYCLES WERE CONDUCTED ON LANDING GEAR TEST ARTICLE (SIMULATOR).

ACCEPTANCE-ACTUATOR EXTEND TIME TEST, PERFORMANCE RECORD TEST INCLUD HIGH PRESSURE STATIC EXTERNAL LEAKAGE TEST, LOW PRESSURE STATIC EXTER LEAKAGE TEST AND DYNAMIC SEAL LEAKAGE TEST. CHECK VALVE IS TESTED AT INDIVIDUAL COMPONENT LEVEL FOR CRACK, RESEAT AND PROPER FLOW. U CLEANLINESS TEST.

OMRSD-TIMED LANDING GEAR EXTENSION EVERY TURNAROUND.

(C) INSPECTION ...

RECEIVING INSPECTION

CERTIFICATION RECORDS AND CERTIFIED TEST REPORTS ARE MAINTAI CERTIFYING MATERIAL AND PHYSICAL PROPERTIES.

CONTAMINATION CONTROL

SUPPLIER TEST STAND FLUID PARTICLE COUNT CHECKED TWICE A DAY, W APPLICABLE. FLUID CONTAMINATION PARTICLE COUNT CONDUCTED PRIOR TO A AFTER ATP. A FLUID SAMPLE IS DRAWN TO VERIFY FLUID CLEANLINESS. CONTAMINATED, ACTUATOR IS CYCLED AND FLUSHED UNTIL CONFIRMATION ATTAINED.

CRITICAL PROCESSES

HEAT TREATMENT IS VERIFIED BY INSPECTION.

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INSPECTION VERIFIES THAT SPRINGS ARE BOTH MAGNETIC PARTICLE AND PENETR INSPECTED. OTHER DETAIL PARTS ARE MAGNETIC PARTICLE OR PENETR INSPECTED, DEPENDING ON THE ALLOY, AND VERIFIED BY INSPECTION.

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SUBSYSTEM : LANDING DECELERATION FMEA NO 02-1F -G09-CC-J01 REV: 06/27/8

ASSEMBLY/INSTALLATION

INSPECTION OF SPRING DIMENSIONS AND OTHER DIMENSIONS IS VERIFIED. QUALITY CONTROL WITNESSES SEAL AND BACKUP RING INSTALLATION. O-RI GROOVES AND SEAL FACES INSPECTED FOR PROPER FINISH. ALL SEALS INSPECT PRIOR TO INSTALLATION. COMPONENT PARTS VERIFIED UNDAMAGED PRIOR CLEANING AND PACKAGING.

TESTING

ATP IS VERIFIED BY INSPECTION.

HANDLING/PACKAGING

PARTS PROTECTION TO PRECLUDE CONTAMINATION DURING SHIPMENT IS VERIFIED INSPECTION.

- (D) FAILURE HISTORY
 THERE IS NO HISTORY OF FAILURE FOR THIS FAILURE MODE.
- (E) OPERATIONAL USE

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